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#### **ORIGINAL ARTICLE**



# Low-temperature oxidation-induced crack healing in Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C-Al<sub>2</sub>O<sub>3</sub> composites

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#### Abstract

The oxidation-induced crack healing of an Al<sub>2</sub>O<sub>3</sub> composite loaded with various volume fractions of Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C repair filler particles was investigated by annealing in air at a relatively low temperature of 700°C. After annealing a composite with 20 vol.% repair fillers (with a particle size of ~5.6 µm) for 48 hours, artificial indentation cracks prepared on the surface, as well as pores near the surface, were completely healed by filling with condensed oxidation products. Additionally, minor fraction of metallic Sn was detected. A complementary study by X-ray diffraction, transmission electron microscopy, scanning electron microscopy, and energy dispersive X-ray spectroscopy revealed that nano-sized oxidation products (SnO<sub>2</sub>, TiO<sub>2</sub>, and α-Al<sub>2</sub>O<sub>3</sub> phase) were formed as major crack-filling species. After healing, the strength recovery of the Al<sub>2</sub>O<sub>3</sub> composites was significantly improved in the composites loaded with more than 10 vol.% repair fillers and achieved 107% at 700 for 48 hours.

#### **KEYWORDS**

Al<sub>2</sub>O<sub>3</sub>-MAX phase composites, oxidation products, oxidation-induced crack healing, strength recovery

#### 1 **INTRODUCTION**

Engineering ceramics being able to repair cracks upon heat treatment have gained increasing attention.<sup>1-3</sup> Crack healing may offer a high potential for improving the reliability and prolongation of the lifetime of ceramic components subjected to mechanical loading at elevated temperatures.<sup>1</sup> Crack healing in ceramics via re-sintering (based on UO<sub>2</sub>,<sup>4,5</sup> Al<sub>2</sub>O<sub>3</sub>,<sup>6,7</sup> ZnO,<sup>8</sup> MgO<sup>9</sup>), as well as oxidation of SiC, Si<sub>3</sub>N<sub>4</sub>, and related composites,<sup>10-14</sup> were reported as major crackhealing mechanisms giving rise for partial or even full recovery of the strength. The oxidation mechanism exhibits a more efficient healing ability than the re-sintering mechanism, because the volume expansion induced by crack surface oxidation can fill the crack gap more effectively. The enhancement of the crack-healing ability of those ceramics that are controlled by re-sintering mechanisms, was

successfully achieved by loading repair fillers, such as SiC particles or whiskers in the ceramic matrix, featuring oxidation-induced healing.<sup>15–22</sup> Several parameters affecting the healing efficiency were investigated, such as healing temperature and time,<sup>15</sup> stress,<sup>18</sup> crack dimension,<sup>19</sup> and oxygen partial pressure,<sup>20</sup> as well as volume fraction and particle size of the repair filler constituent.<sup>21</sup> For example, an enhanced healing ability was observed in Al<sub>2</sub>O<sub>3</sub> composites loaded with submicron-sized SiC particles as repair filler.<sup>21</sup> Decreasing the repair filler particle size from 270 to 30 nm resulted in lowering of the healing temperature from 1300 to 950°C, which was attributed to the activation energy for oxidation, scaling with SiC repair filler particle size.<sup>21</sup> The presence of an activator, such as MoO (0.2 vol.%) in the SiC-Al<sub>2</sub>O<sub>3</sub> composite, was found to accelerate crack healing significantly, while strength recovery was achieved at 1000°C for 1 hour healing period.<sup>22</sup>

Recently, a group of ternary nitrides and carbides (MAX phases) with the general formula  $M_{n+1}AX_n$  (n = 1 to 3), where M is a transition metal, A is an A group element, and X is either carbon or nitrogen,<sup>23</sup> demonstrated interesting healing abilities. MAX phases containing Al, such as Ti<sub>3</sub>AlC<sub>2</sub>,<sup>24</sup>  $Ti_2AlC$ , <sup>25–27</sup> and  $Cr_2AlC$ , <sup>28</sup> were reported to form a dense layer of  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> filling in the gap. Compared with SiC- and Si<sub>3</sub>N<sub>4</sub>-based composites, larger cracks with a length of up to 7 mm and a crack opening width of up to 5 µm could be fully healed in Ti<sub>3</sub>AlC<sub>2</sub> after heat treatment at 1100°C for 2 hours in air. Even a repeatable crack healing was observed in Ti<sub>2</sub>AlC, indicating that MAX phases offer a multiple crackhealing ability. Furthermore, a Sn-containing MAX phase (Ti<sub>2</sub>SnC) was able to repair millimeter-sized cracks by annealing at a relatively low temperature of 800°C within only 1 hour in air,<sup>29</sup> as well as in vacuum.<sup>30</sup> After healing, the flexural strength<sup>29</sup> and electrical conductivity<sup>29,30</sup> of the damaged material were almost fully recovered and reached the level of the virgin material.  $Ti_2Al_xSn_{(1-x)}C$  MAX phase solid solutions were able to undergo oxidation-induced crack healing in ceramic composites at temperatures even below 1000°C.<sup>29-33</sup> The fracture strength of Al<sub>2</sub>O<sub>3</sub> composites loaded with 20 vol.% of the Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C repair filler containing artificial indent cracks, recovered fully to the level of the virgin material upon isothermal annealing in air atmosphere for 0.5 hours at 900°C.<sup>32</sup> However, the intrinsic healing mechanisms of  $Ti_2Al_xSn_{(1-x)}C-Al_2O_3$  at temperatures even lower than 700°C have not been discovered yet. Thus, the scope of the present work was to examine the crack-healing behavior of Al<sub>2</sub>O<sub>3</sub> composites loaded with Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C repair filler operating at 700°C. An improved distribution of the repair filler in the Al<sub>2</sub>O<sub>3</sub> matrix composites were obtained by reducing the particle size of repair filler. The healing efficiency was correlated with the oxidation and microstructure analysis of the healed zone of Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C-Al<sub>2</sub>O<sub>3</sub> composites.

# 2 | EXPERIMENTAL DETAILS

The Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C solid solution was synthesized from a reactant powder mixture, consisting of Ti, Al, Sn, and TiC (2  $\mu$ m, 99% purity) with a molar composition corresponding to Ti-0.5Sn-0.5Al-0.9TiC. The annealing was conducted for 1 hour in vacuum at 1400°C. The reaction product was milled for 12 hours (Attrition mill with a ball-to-materials ratio of 2:1), resulting in a mean particle size of 5.6  $\mu$ m (Mastersizer 2000, Malvern Instruments, UK). The Al<sub>2</sub>O<sub>3</sub> matrix (AKP-53; Sumitomo Chemical, Japan) loaded with Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C composites with repair filler fractions of 5, 10 and 20 vol.% were sintered at 1350°C for 4 hours in Ar atmosphere (Heraeus Holding GmbH, Hanau, Germany) by applying a heating rate of 5°C/min. Further details of the manufacturing process are given in Ref<sup>30,31</sup>.

The Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C–Al<sub>2</sub>O<sub>3</sub> composite, dedicated for mechanical investigation, was polished to 1 µm surface finish using a diamond suspension and cut into bar specimens with dimensions of  $2.5 \times 2.0 \times 28$  mm<sup>3</sup>. The density of the materials was measured according to the Archimedes method. Surface cracks were generated by means of Vickers indentation, applying a constant load of 100 N (HV10) for 10 seconds (Zwick, Ulm, Germany). Oxidation-induced crack healing was carried out in an oxidation furnace (Linn High Term GmbH, Eschenfelden, Germany) at 700°C for 48 and 96 hours. The weight change of Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C–Al<sub>2</sub>O<sub>3</sub> composites caused by oxidation reaction was recorded at 700 and 900°C for 12 hours by a thermal balance, applying a heating rate of 5°C/min (STA 429; Netzsch, Selb, Germany).

The phase composition prior and after the heat treatment was analyzed by X-Ray diffraction (XRD, Kristalloflex; Siemens AG, Mannheim, Germany) operated with monochromated Cu- $K_{\alpha}$  radiation. The composite microstructure and the indent crack morphology were analyzed by scanning electron microscopy (SEM, Helios NanoLab 600i FIB Workstation; FEI, Eindhoven, the Netherlands). After the healing treatment, the reaction products filling the crack space were examined by field emission SEM (FE-SEM) coupled with a focus ion beam (FIB, Helios NanoLab 600i FIB Workstation, FEI) system and energy dispersive X-ray spectroscopy (EDXS; Oxford Instruments INCA, Oxford, UK). Thin-section specimens were prepared by FIB, which is equipped with a Ga<sup>+</sup>-ion source and operated at 30 kV. A protective platinum layer  $(30 \times 2 \times 5 \text{ } \mu\text{m}^3)$  was deposited on the surface area of the selected healed zone. The lamella was finally thinned to 50-100 nm with a very fine ion beam current. Transmission electron microscopy (TEM) was performed with a Philips CM30 TWIN/STEM and Philips CM300 UltraTWIN (both from FEI Company), both operated at 300 kV acceleration voltage. TEM images and electron diffraction patterns were recorded using a charged couple device (CCD) camera from TVIPS (Tietz Video and Processing Systems GmbH, Gauting, Germany) with an image size of  $1024 \times 1024$  pixels (at the CM30 TEM) and  $2048 \times 2048$ pixels (at the CM300 TEM), respectively. The processing of the TEM images and diffraction patterns was performed with the free available software ImageJ (Version 1.48r) and the commercially available software DigitalMicrograph<sup>TM</sup>. The evaluation of the electron diffraction patterns was performed by simulating the experimental diffraction patterns with the software JEMS (java version 3.5505U2010) and using the inorganic crystal structure database (ICSD) files for Al<sub>2</sub>O<sub>3</sub> (#10425), SnO<sub>2</sub> (#9163), and TiO<sub>2</sub> (#9161).

The Gibbs free energy of various oxidation products, such as  $Al_2O_3$ ,  $SnO_2$ , and  $TiO_2$ , was calculated by thermodynamic calculations using the FactSage 7.1 software.<sup>34</sup> The thermodynamic data of oxidation products  $Al_2O_3$ ,  $SnO_2$ , and  $TiO_2$  were taken from the FToxide database and the thermodynamic data of metallic Al and Sn and Ti were taken from the Bins database.<sup>34</sup>

Filling of the crack space with condensed oxidation products causes the strength of the indented samples to recover. The flexural strength of the virgin sample,  $\sigma_0$ , indented sample,  $\sigma_{indent}$ , as well as healed samples after thermal treatment,  $\sigma_{heal}$ , were measured by three-point bending (5565; Instron Deutschland GmbH, Pfungstadt, Germany) using ASTM Standard (C1161-18), by applying a lower support span of 20 mm and a crosshead speed of 0.5 mm/min.

# **3** | **RESULTS AND DISCUSSION**

### 3.1 | Composite microstructure

Figure 1 displays the XRD pattern of the sintered Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C-Al<sub>2</sub>O<sub>3</sub> composites loaded with 20 vol.% repair fillers. Only  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> and Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C solid solution were detected as crystal phases, indicating that the Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C repair filler did not suffer from thermal degradation during the high-temperature sintering process. Figure 2 shows back scattering SEM images of sintered Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C-Al<sub>2</sub>O<sub>3</sub> composites containing 5, 10, and 20 vol.% repair fillers. The MAX phase repair filler particles were uniformly dispersed in the alumina matrix. In accordance with the increasing volume fraction of the repair filler, the mean repair filler inter-particle distance,  $\lambda_{MAX}$ , calculated according to Ref. <sup>35</sup>, decreased from  $\lambda_{MAX} \approx 2.3 \,\mu m$ (5 vol.%) to ~1.3  $\mu$ m (20 vol.%), respectively (see Table 1). The repair filler inter-particle distances are significantly smaller compared to the lengths of the artificial cracks emanating from the load indents ( $c_{indent} = 280 \sim 330 \ \mu m$ ). Figure 3 shows a typical microstructure of a crack emanating from the tip of the Vickers indentation in the  $Al_2O_3$ 



**FIGURE 1** X-Ray diffraction pattern of the synthesized  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composites containing 20 vol.% repair fillers with a particle size of 5.6 µm



**FIGURE 2** Scanning electron microscopy micrograph of  $Al_2O_3$  composites sintered at 1350°C for 4 h in Ar atmosphere containing different volume fraction of  $Ti_2Al_{0.5}Sn_{0.5}C$  MAX phase repair filler (A) 5 vol.%, (B) 10 vol.% (C) 20 vol.%

<b>FABLE 1</b>	Microstructure parameters of the prepared alumina
composite spec	timens

Vol.%	5	10	20
Porosity (%)	2.1	2.8	7.6
$d_{\max}$ (µm)	$4.8 \pm 2.2$	$4.5 \pm 1.4$	$4.5 \pm 1.5$
$\lambda_{\mathrm{MAX}}\left(\mu m\right)$	$2.3 \pm 1.3$	$1.7 \pm 1.2$	1.3 ± 1.1
Indent crack length (µm)	345 ± 9	336 ± 4	328 ± 8
Indent crack width (µm)	$0.43 \pm 0.04$	$0.46 \pm 0.05$	$0.32\pm0.06$



**FIGURE 3** Vickers indentation induced crack formation in the Al<sub>2</sub>O<sub>3</sub> composites containing 10 vol.% repair fillers with different particle size

composite containing 10 vol.% repair fillers. The repair filler particles are homogenously distributed around the crack path. A small prolongation of the crack length of 345  $\mu$ m observed on composites with a minimum repair filler fraction of 5 vol.% may be attributed to a reduced toughness compared to an enhanced toughness at high particle loading fractions.

# 3.2 | Oxidation-induced crack healing

Annealing of the  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composites in air triggered a sequence of oxidation reactions of the  $Ti_2Al_{0.5}Sn_{0.5}C$  repair filler, which resulted in the formation of  $Al_2O_3$ ,  $SnO_2$ , and  $TiO_2$  oxidation products (Figure 4). Although, the Gibbs free energy for  $Al_2O_3$  formation is the lowest among the formed oxides (Figure 5), newly formed  $Al_2O_3$  could not be detected by XRD due to overlapping  $Al_2O_3$  matrix peaks. Metallic Sn was detected, which is likely to occur deeply in the crack where the local oxygen concentration is too low for tin oxide formation.<sup>32</sup>

Figure 6 shows the weight increase of the composite loaded with 20 vol.%  $Ti_2Al_{0.5}Sn_{0.5}C$  during the thermogravimetric measurement at 700 and 900°C under same



**FIGURE 4** X-Ray diffraction pattern of  $Al_2O_3$  composites containing 20 vol.%  $Ti_2Al_{0.5}Sn_{0.5}C$  repair filler after heating at 700°C for 48 h

heating conditions. Oxidation reactions, indicated by a weight increase, started at about 450°C, which was in agreement with the oxidation behavior measured on the single phase Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C powders.<sup>36</sup> While at 900°C a rapid reaction causes the weight gain to raise to 4.25 after 30 minutes, the oxidation reaction at 700°C proceeds much slower and reaches 3.3% after a holding period of 12 hours. Furthermore, time scaling changes from a linear relation at 900°C to a parabolic relation at 700°C. Figure 7 shows the microstructure of the healed crack after annealing at 700°C for an elongated period of 48 hours. Crack bridge formation and partial crack filling was observed on the composite loaded with 10 vol.% repair fillers, whereas complete crack filling occurred in the 20 vol.% specimen. EDXS mapping analysis confirmed the presence of TiO<sub>2</sub>, SnO<sub>2</sub> and Al<sub>2</sub>O<sub>3</sub> as major oxidation products, as well as a minor fraction of metallic Sn.



**FIGURE 5** Standard Gibbs free energy of various oxidation reactions of M and A elements from the  $Ti_2Al_{0.5}Sn_{0.5}C$  MAX phase



**FIGURE 6** Weight gain upon oxidation annealing of the  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composite (20 vol.%) in air at 700 and 900°C for 12 h



Figure 8A is a representative bright field (BF) TEM image showing the  $Al_2O_3$  matrix and the healed area. As indicated in Figure 8A by the white dashed line, an



**FIGURE 7** Scanning electron microscopy (SEM) micrographs of cracks after annealing at 700°C for 48 h (A) 10 vol.%  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composites, (B) 20 vol.%  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$ composites, (C) a cross section of the healed crack in the 20 vol.%  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composites prepared by focus ion beam and corresponding SEM-energy dispersive X-ray spectroscopy elemental mapping analysis

interface clearly separates the Al<sub>2</sub>O<sub>3</sub> matrix form the healed zone. Selected area electron diffraction (SAED) confirms the presence of single-crystalline orthorhombic  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> (see Figure 8B). The healed area, as well as the interface between the  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> matrix and the healed zone, are examined in more details by high-resolution TEM (HR-TEM). The  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> matrix exhibits a typical fringe periodicity of 0.21 nm, which fits well with the theoretical lattice spacing of  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> (113) planes (orthorhombic crystal structure, ICSD #10425). The healed area consists of nanocrystallites (which is evident from Bragg-contrasts in Figure 8A) with a random orientation (see HR-TEM images in Figure 8C,D). The observed spacing of 0.26 nm can be assigned to (104) lattice planes of orthorhombic  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> (ICSD #10425), while the spacing values of 0.17 and 0.18 nm can be assigned to (211) lattice planes of tetragonal TiO<sub>2</sub> (ICSD #9161) and (211) lattice planes of tetragonal SnO<sub>2</sub> (ICSD #9163), respectively. However, since the lattice spacing values of 0.17 and 0.18 nm are very close to each other, a clear differentiation and their assignment to specific lattice planes is problematic. A misinterpretation of the lattice planes can affect conclusions about the phases, which in this particular case also have an identical (tetragonal) crystal structure with similar lattice parameters. However, in relation and agreement with XRD and SEM-EDXS mapping results, we conclude that both phase, tetragonal TiO<sub>2</sub> and SnO<sub>2</sub>, are present (and co-existing with  $\alpha$ -Al<sub>2</sub>O<sub>3</sub>) in the healed zone. Furthermore, amorphous regions are detected at the interface between  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> and the healed zone, as can be seen in Figure 8C. Also in the interior of the healed zone, an amorphous phase is observed between the nanocrystallites (see Figure 8D). There are two possible reasons for these observations:

1. The annealing temperature of 700°C was relatively low. The temperature may be insufficient for complete crystallization and growth of oxidation products (TiO<sub>2</sub> + SnO<sub>2</sub> +  $\alpha$ -Al<sub>2</sub>O<sub>3</sub>), which is in good agreement with the HR-TEM results from the healed area (Figure 8C,D). According to previous results, a large amount of crystalline TiO<sub>2</sub> (>5 µm) and  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> (<1 µm) can be expected after annealing at 900°C for 1 hour or above 1000°C.<sup>37,38</sup>



**FIGURE 8** Transmission electron microscopy (TEM) analysis of the 20 vol.% Ti<sub>2</sub>Al<sub>0.5</sub>Sn<sub>0.5</sub>C–Al<sub>2</sub>O<sub>3</sub> composite after heat treatment at 700°C for 48 h (A) bright field-TEM image of the Al<sub>2</sub>O<sub>3</sub> matrix, separated by an interface from the healed area; (B) selected area electron diffraction pattern (valid for the [8 10 1] zone axis) of the orthorhombic  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> (ICSD #10425); (C) and (D) are high-resolution TEM images, confirming the presence of a single-crystalline  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> phase, which is separated by an interface from the healed zone, composed of  $\alpha$ -Al<sub>2</sub>O<sub>3</sub>, TiO<sub>2</sub>, and SnO<sub>2</sub> nanocrystals

 Ga<sup>+</sup>-ion beam-induced damage during FIB preparation and possible electron beam (e-beam)-induced damage during the TEM analysis may also cause an amorphization of the specimen. Ion-beam-induced damage is often observed during the thinning procedure of the outer regions of TEM samples.<sup>39</sup>

# 3.3 | Strength recovery

Figure 9 shows the variation of the bending strength prior and after healing treatment at 700°C for 48 and 96 hours. Indentation caused the virgin strength to decrease for more than 50%. After annealing in air, the recovery of strength up to the level of the virgin material (and even slightly higher due to the healing of small surface cracks and pores) can be observed for the composites loaded with 10 and 20 vol.% repair fillers. Since the mean coefficients of thermal expansion of the crack-filling oxide reaction products (TiO<sub>2</sub> [rutile]:  $8.4 \times 10^{-6}$ /K)<sup>40</sup> and SnO<sub>2</sub>:  $3.9 \times 10^{-6}$ /K)<sup>41</sup> are smaller than the one of the  $\alpha$ -Al<sub>2</sub>O<sub>3</sub> matrix material (8.4 × 10<sup>-6</sup>/K),<sup>38</sup> compressive stresses are likely to be generated at the crack-matrix interface upon cooling, which tend to increase the crack growth resistance.<sup>32</sup> In addition to healing of the artificial indent cracks, closure of the residual porosity in the composite (2%-7%) might also contribute to the strength recovery.



**FIGURE 9** Virgin strength, indented strength and recovered strength of the  $Ti_2Al_{0.5}Sn_{0.5}C-Al_2O_3$  composites with varied volume fraction of  $Ti_2Al_{0.5}Sn_{0.5}C$  repair filler

# 4 | CONCLUSION

The oxidation-induced crack healing of a  $Al_2O_3$  composite loaded with the  $Ti_2Al_{0.5}Sn_{0.5}C$  MAX phase repair filler was examined at 700°C. The formation of nano-sized TiO<sub>2</sub>



(rutile) +  $\text{SnO}_2$  +  $\alpha$ -Al<sub>2</sub>O<sub>3</sub>, as well as a minor fraction of metallic Sn as the crack-filling material, is observed. Restoration of the solid contact between the crack surfaces triggers complete recovery of the compromised strength at repair filler fractions exceeding 10 vol.%.

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